

BENDING ANALYSIS OF STRAIN-SOFTENING BEAMS

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1. INTRODUCTION

It is well known that materials such as concrete, soil and rock exhibit a strain softening behaviour after reaching a peak stress in compressive tests. For example, the stress-strain curve of a concrete in standard cylinder tests rises to a strain about 0.25% and afterward falls gradually to a strain about 0.35% when a crushing failure occurs. Since a strain softening is a phenomenon of physical instability, in an analysis of the structures made of strain softening materials, a question about stability and uniqueness of solution arises.

For example, in a finite element displacement analysis, a final matrix equation between the unknown nodal displacements or forces, \mathbf{P} , and the imposed nodal forces or displacements, \mathbf{u} , under consideration of prescribed boundary conditions is given as

$$\mathbf{P} = \mathbf{K}\mathbf{u} \quad \dots\dots\dots(1)$$

If the overall stiffness matrix \mathbf{K} is positive definite, a solution of Eq. (1) can uniquely be determined from

$$\mathbf{u} = \mathbf{K}^{-1}\mathbf{P} \quad \dots\dots\dots(2)$$

But, in the case where \mathbf{K} is not positive definite, Eq. (2) does not give a stable equilibrium solution making a total potential energy of system minimum, because the function of strain energy loses its convexity. Concerning this problem, a stability condition of strain softening structures was discussed by Prevost & Hoeg¹⁾ for a three dimensional body and Maier²⁾ for a beam.

On the other hand, for reasons of the urgent necessity of solving some engineering problems of soil foundations, rock foundations and concrete structures, various numerical analyses of strain softening structures have been carried out^{3)-6),9),10)}, though a question mentioned above remains open. To determine load carrying capacities of such structures is one of the most important engineering problems. If regions

exhibiting strain softening are relatively small and are confined by strong surrounding elastic constraints, as often encountered in practical problems, the structures may withstand still higher loads as a stable structure after the strain energy stored within the regions of strain softening was perfectly released. For an analysis of such problems, an appropriate treatment of local instability due to strain softening becomes very important.

From such a viewpoint, this paper has an attempt to obtain a method of analysis of elastic-plastic-strain softening beams. The method is developed with the aid of a minimum principle of strain energy-increment, using a matrix displacement method. A detailed procedure of numerical calculation is described through a simple example of a two-span beam.

2. MINIMUM PRINCIPLE OF STRAIN ENERGY-INCREMENT

A minimum principle of energy says that an equilibrating solution exists at the point making a total potential of system minimum among any kinematically admissible displacement fields. Such a potential is expressed, in the absence of body forces, as follows:

$$\pi = \int_V W(\epsilon_{ij}) dV - \int_{S_i} T_j u_j dS \quad \dots\dots\dots(3)$$

where ϵ_{ij} is strain tensor, $W(\epsilon_{ij})$ is strain energy, T_j , u_j are surface force and displacement, S_i is the boundary surface subjected to T_j , and V is the entire body considered here. When the body is subjected to a small change of the external agency from (T_j, u_j) to $(T_j + dT_j, u_j + du_j)$, Eq. (3) yields

$$\pi + d\pi = \int_V W(\epsilon_{ij} + d\epsilon_{ij}) dV - \int_{S_i} (T_j + dT_j)(u_j + du_j) dS \quad \dots\dots\dots(4)$$

From Eqs. (3) and (4), we obtain

$$d\pi = \int_V \left[\frac{\partial W}{\partial \epsilon_{ij}} d\epsilon_{ij} + f(d\epsilon_{ij}) \right] dV - \int_{S_i} [T_j du_j + dT_j(u_j + du_j)] dS \quad \dots\dots\dots(5)$$

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where $f(d\varepsilon_{ij})$ is the strain energy being of higher order than the second order.

Using the stationary condition, $\delta\pi=0$, minimizing of $d\pi$ in a displacement field is reduced to that of $d\pi'$ as follows:

$$d\pi' = \int_V f(d\varepsilon_{ij}) dV - \int_{S_t} dT_j(u_j + du_j) dS \quad \dots\dots\dots(6)$$

Generally, for a numerical analysis of strain softening structures, the use of controlling a displacement-increment on the surface S_t is more convenient than that of controlling load-increment on it, because of relaxing a numerical instability due to a strain softening effect. Here, introducing an incremental multiplier of load denoted by

$$d\lambda = dT_j/T_j^0 \quad \dots\dots\dots(7)$$

where T_j^0 is a normalized load distribution, a controlled displacement-increment is given as

$$dL = \int_{S_t} T_j^0 du_j dS \quad \dots\dots\dots(8)$$

Thus, minimizing of $d\pi'$ in Eq. (6) is reduced to

$$\left. \begin{array}{l} \text{minimize} \\ \int_V f(d\varepsilon_{ij}) dV \\ \text{subject to} \\ \int_{S_t} T_j^0 du_j dS = dL \end{array} \right\} \dots\dots\dots(9)$$

Finally, using a solution $d\varepsilon_{ij}^*$ of Prob. (9), from the principle of virtual work, we obtain

$$d\lambda = \frac{1}{dL} \int_V \frac{\partial f}{\partial (d\varepsilon_{ij})} \Big|_{d\varepsilon_{ij} = d\varepsilon_{ij}^*} \cdot d\varepsilon_{ij}^* dV \quad \dots\dots(10)$$

3. BENDING ANALYSIS OF BEAMS

Let us consider a beam with a cross sectional property whose moment-curvature relation is given by a bilinear curve shown in Fig. 1. But, when such strain softening characteristics of moments decreasing with curvature are directly used, we encounter the troublesome situation that curvatures at the points reaching to yield moment can no longer increase and in consequence only elastic recovery can occur. This contradiction was pointed out by Wood⁷⁾. And Wood⁷⁾ and Barnard & Johnson⁸⁾ indicated that it can be removed by introducing a small but finite region of highly localized strain softening at the points of peak moment, which was named the discontinuity line.

Here, introducing a discontinuity length denoted by Δl , we can replace the cross sectional characteristics of Fig. 1 by a plastic-strain softening hinge as shown in Fig. 2. Thus, putting

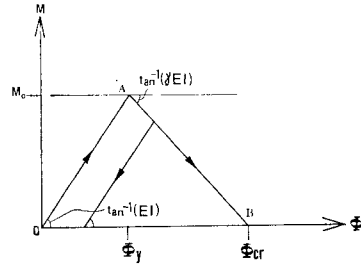


Fig. 1 Moment-curvature relation.

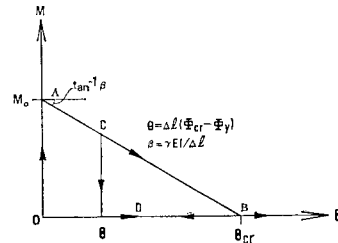


Fig. 2 Plastic-strain softening hinge.

$$dW = \int_V f(d\varepsilon_{ij}) dV,$$

Prob. (9) reduced, in a finite element analysis of beam bending, to

$$\left. \begin{array}{l} \text{minimize} \\ dW = \frac{1}{2} \sum_i d\delta_i^t \mathbf{K}_i d\delta_i - \frac{1}{2} \sum_i \beta_j \Delta\theta_j^2 \\ \text{subject to} \\ \mathbf{a} d\delta + \mathbf{b} \Delta\theta = \mathbf{0} \\ \mathbf{P}_0^t d\delta = dL, \quad \Delta\theta \geq \mathbf{0} \end{array} \right\} \dots\dots(11)$$

where $d\delta_i$ is displacement-increments at the extremities of an element i ; \mathbf{K}_i is element stiffness matrix; $\Delta\theta_j$ is plastic rotation-increment at a hinge j , which takes a positive value when its direction coincides with that of moment; β_j is a coefficient of linear strain softening; $d\delta^t = [d\delta_1 d\delta_2 \dots]$; $\Delta\theta^t = [\Delta\theta_1 \Delta\theta_2 \dots]$; and \mathbf{P}_0 is a normalized load vector. The first equality constraint in Prob. (11) represents a compatibility between the plastic rotation-increment and the slope-increments at the adjacent extremities. Namely, in the case where $\Delta\theta = \mathbf{0}$ identically, Prob. (11) is reduced to a purely elastic problem.

Further, to eliminate the free variables $d\delta$ from our problem, the following lagrangian function is introduced:

$$F = dW + \mu^t (\mathbf{a} d\delta + \mathbf{b} \Delta\theta) + \mu_0 (\mathbf{P}_0^t d\delta - dL) \quad \dots\dots\dots(12)$$

where μ , μ_0 are Lagrange multipliers. Thus, from the stationary condition; $\partial F / \partial d\delta = \mathbf{0}$, $\partial F /$

$\partial\mu=0$, and $\partial F/\mu_0=0$; Prob. (11) can be reduced to the following form:

$$\left. \begin{aligned} &\text{minimize} \\ &d\bar{W} = -dLA^t\Delta\theta + \Delta\theta^t D\Delta\theta \\ &\text{subject to} \\ &\Delta\theta \geq 0 \end{aligned} \right\} \dots\dots\dots(13)$$

Next, consider a solution of Prob. (13).

In Case 1 where D is positive definite, namely $\chi^t D \chi > 0$ for $0 < |\chi| < \infty$, the function of $d\bar{W}$ has a convexity. Then, there exists a unique solution, $\Delta\theta^*$, satisfying the following Kuhn-Tucker condition:

$$\left. \begin{aligned} &2D\Delta\theta^* - dLA - v = 0 \\ &\Delta\theta^* \geq 0, \quad v \geq 0, \quad \Delta\theta^{*t} v \geq 0 \end{aligned} \right\} \dots\dots\dots(14)$$

where v is slack variables.

For this case, the methods of quadratic programming such as Wolfe's method or Beal's method⁽¹¹⁾ can be utilized.

In Case 2 where D is copositive, namely $\chi^t D \chi > 0$ for only $\chi > 0$, the function $d\bar{W}$ has no convexity. However, a solution satisfying the condition (14) exists, because the function, $d\bar{W}$, is bounded in low. Namely, there is a stable equilibrium point in a local sense. For this case, Ritter's method⁽¹²⁾ for nonlinear programming may be utilized. A stability condition given by Maier⁽²⁾ is related with this case.

In Case 3 where D is neither positive definite nor copositive, Prob. (13) has no solution, because $d\bar{W}$ is unbounded in low. Namely, there is no stable equilibrium point in the strain softening domain A-B in Fig. 2. Then, in this case, a jumping from the current state to a point in the domain 0-B of no strength in Fig. 2 occurs after the elastic strain energy stored in the hinges was perfectly released.

Detailed procedures of numerical calculation related with the three cases mentioned above will be described through a simple example of a two-span beam in Fig. 3. According to an ordinary matrix displacement method, the beam is discretized by the three elements shown in Fig. 3. A plastic-strain softening hinge is possible at the points 1, 2, and 3. Thus, Problem (11) under an imposed deflection-increment du_2 at the loading point 2 is expressed as

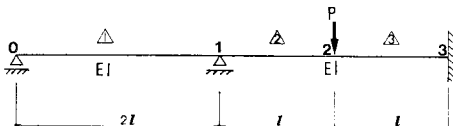


Fig. 3 Example of a two-span beam.

minimize

$$\left. \begin{aligned} \frac{3l}{2EI} dW = &\frac{9}{8} \varphi_{1b}^2 + (\varphi_{2a} - \varphi_{2b}) \left(2\varphi_{2a} + \varphi_{2b} - 3 \frac{du_2}{l} \right) \\ &+ \left(2\varphi_{2b} + \varphi_{2a} - 3 \frac{du_2}{l} \right)^2 \\ &+ (\varphi_{3a} - \varphi_{3b}) \left(2\varphi_{3a} + \varphi_{3b} + 3 \frac{du_2}{l} \right) \\ &+ \left(2\varphi_{3b} + \varphi_{3a} + 3 \frac{du_2}{l} \right)^2 \\ &- \beta_1' \Delta\theta_1^2 - \beta_2' \Delta\theta_2^2 - \beta_3' \Delta\theta_3^2 \end{aligned} \right\}$$

subject to

$$\begin{aligned} \varphi_{2a} - \varphi_{1b} - \Delta\theta_1 &= 0, & \varphi_{2a} - \varphi_{3a} - \Delta\theta_2 &= 0, \\ \varphi_{3b} + \Delta\theta_3 &= 0, & P_0 \frac{du_2}{l} - dL &= 0, \\ \Delta\theta_1 \geq 0, & \Delta\theta_2 \geq 0, & \Delta\theta_3 \geq 0. \end{aligned}$$

.....(15)

where φ_{ia} and φ_{ib} ($i=1, 2, 3$) are slope-increment at the left and the right extremities of an element i , respectively, the positive sign of which is taken in the clockwise; $\Delta\theta_j$ ($j=1, 2, 3$) is plastic rotation-increment at a plastic-strain softening hinge j ; and $\beta_j' = 3l\beta_j/4EI$ ($i=1, 2, 3$), in which EI is flexural rigidity.

Introducing the lagrangian function (12) given as

$$\begin{aligned} F = &dW' + \mu_1(\varphi_{2a} - \varphi_{1b} - \Delta\theta_1) \\ &+ \mu_2(\varphi_{2b} - \varphi_{3a} - \Delta\theta_2) \\ &+ \mu_3(\varphi_{3b} + \Delta\theta_3) + \mu_0(P_0 du_2/l - dL), \end{aligned} \dots\dots\dots(16)$$

where $dW' = 3l dW / (2EI)$,

and making its stationary condition; $\partial F / \partial \varphi_{ia} = 0$, $\partial F / \partial \varphi_{ib} = 0$, $i=1, 2, 3$, and $\partial F / \partial \mu_i = 0$, $i=1, 2, 3$; we obtain the following relations:

$$\begin{bmatrix} \varphi_{1b} \\ \varphi_{2a} \\ \varphi_{2b} \\ \varphi_{3a} \\ \varphi_{3b} \end{bmatrix} = \begin{bmatrix} 1.2 \\ 1.2 \\ -0.3 \\ -0.3 \\ 0 \end{bmatrix} \frac{dL}{P_0} + \begin{bmatrix} -0.7 & -0.2 & -0.1 \\ 0.3 & -0.2 & -0.1 \\ -0.075 & 0.55 & 0.275 \\ -0.075 & -0.45 & 0.275 \\ 0 & 0 & -1.0 \end{bmatrix} \begin{bmatrix} \Delta\theta_1 \\ \Delta\theta_2 \\ \Delta\theta_3 \end{bmatrix} \dots\dots\dots(17)$$

where $dL/P_0 = du_2/l$.

Thus, the elimination of φ_{ia} , φ_{ib} from Problem (15), using Eq. (17), gives the explicit form of Problem (13) with the following coefficient matrices:

$$A^t = (1/P_0) \cdot [2.7 \quad 7.2 \quad 8.1]$$

$$D = \begin{bmatrix} 0.7875 - \beta_1' & 0.225 & 0.1125 \\ 0.225 & 1.35 - \beta_2' & 0.675 \\ 0.1125 & 0.675 & 2.5875 - \beta_3' \end{bmatrix} \dots\dots\dots (18)$$

A method of solution for this problem is described below. The following notations are used here: \bar{D}_{ii} is a submatrix derived by deleting i column and i row from D , \bar{A}_i is a submatrix derived by deleting i row from A , and d_{ii} is a diagonal element of D .

Case 1; $|D| > 0$, $|\bar{D}_{ii}| > 0$ and $d_{ii} > 0$, $i = 1, 2, 3$. For this case, an iteration technique familiar in an ordinary elastic-plastic incremental analysis is available, even though an algorithm of quadratic programming is not used. Namely, solution; $\Delta\theta = 1/2 \cdot dL \cdot D^{-1} \cdot A$; is first obtained. If all elements of $\Delta\theta$ are positive or zero, they are the required solution satisfying the condition (14). But, if $\Delta\theta_{i^*}$, $i^* \in i = 1, 2, 3$, is negative, then $\Delta\theta_{i^*}$ is put to zero, and D and A are replaced by $\bar{D}_{i^*i^*}$ and \bar{A}_{i^*} , respectively. Thus, repetition of this procedure yields the required solution satisfying the condition (14). An example of solution

is indicated with the contour of $d\bar{W}$ in Fig. 4 for a problem where plastic-strain softening hinges occurred at the points 2 and 3.

Case 2; $|D| < 0$ or $|\bar{D}_{ii}| < 0$ and $d_{ii} > 0$. Since all of nondiagonal elements of D are positive, D is copositive, if and only if all the diagonal elements d_{ii} are positive. In this case, a Kuhn-Tucker point is necessarily on the constraint surface, $\Delta\theta_{i^*} = 0$. Then, finding i^* , $i^* \in i$, where $\bar{D}_{i^*i^*}$ becomes positive definite, we can obtain the required solution through a similar procedure to that of Case 1. An example of solution in Case 2 is indicated with the contour of $d\bar{W}$ in Fig. 5 for a problem including the hinges at the points 2 and 3.

Case 3; $d_{ii} \geq 0$. The function $d\bar{W}$ is unbounded in low, as seen in the contour of Fig. 6 for an example problem. In this case, a strain energy stored in a hinge i^* where $d_{i^*i^*} < 0$ is abruptly released, and therefore the function of strain energy-increment is reduced to

$$d\bar{W} = \bar{M}_{i^*} \Delta\theta_{i^*} - dL A' \Delta\theta + \Delta\theta^c D |_{\beta'_{i^*} = 0} \Delta\theta \dots\dots\dots (19)$$

where $\bar{M}_{i^*} = 3lM_{i^*}/(2EI)$, M_{i^*} is moment at the hinge i^* in the current state. The reduced matrix with $\beta'_{i^*} = 0$ in Eq. (19) becomes positive definite or copositive, then our problem also returns to Case 1 or Case 2. But, in this case, the solution $\Delta\theta$ is not zero even when the imposed displacement-increment dL vanishes, thus a jumping from a plastic hinge to a freely rotating hinge with no strength results, such as a skip from the point C to the point D in Fig. 2.

4. NUMERICAL CALCULATION

Relations between load or bending moment and displacement in the two-span beam with a uniform flexural rigidity, as shown in Fig. 3, are illustrated in Figs. 7, 8, and 9. In these figures;

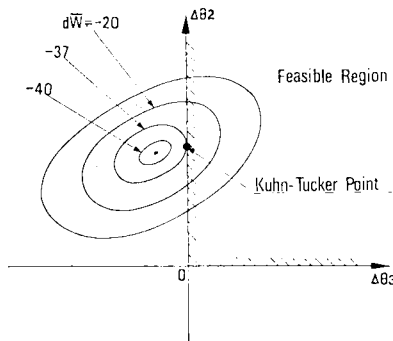


Fig. 4 Contour of $d\bar{W}$ and its minimum point in Case 1 with $\beta_2' = 1.0$ and $\beta_3' = 0.5$, under $dL/P_0 = du_2/l = 1.0$.

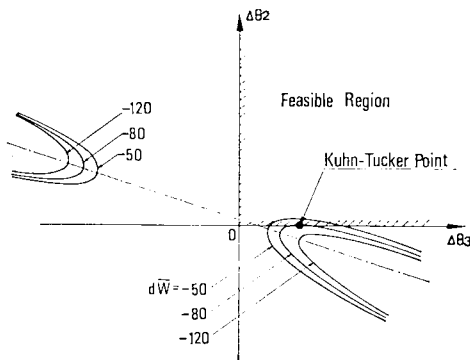


Fig. 5 Contour of $d\bar{W}$ and its minimum point in Case 2 with $\beta_2' = 1.2$ and $\beta_3' = 0.5$, under $dL/P_0 = du_2/l = 1.0$.

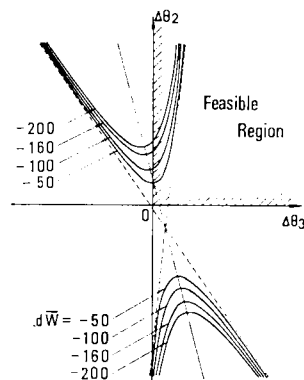


Fig. 6 Contour of $d\bar{W}$ in Case 3 with $\beta_2' = 1.5$ and $\beta_3' = 0.5$, under $dL/P_0 = du_2/l = 1.0$.

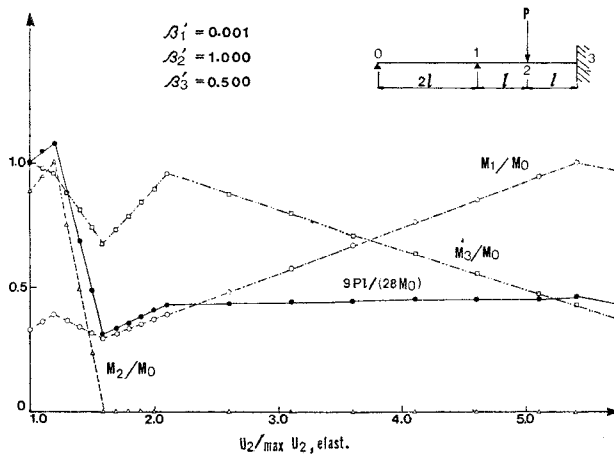


Fig. 7 Relations between bending moments or load and deflection for the problem including only Case 1.

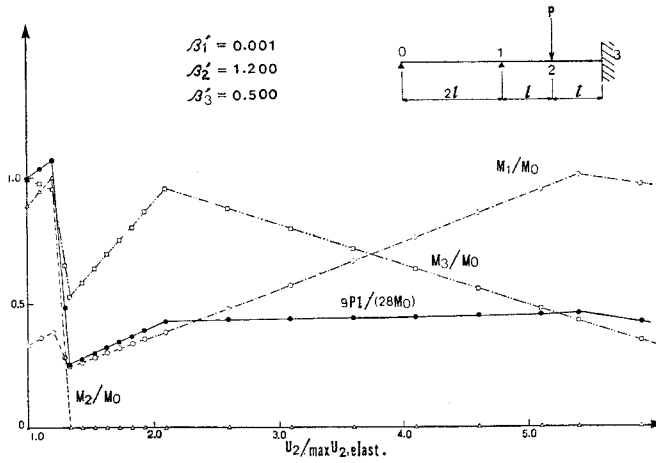


Fig. 8 Relations between bending moments or load and deflection for the problem including both Case 1 and Case 2.

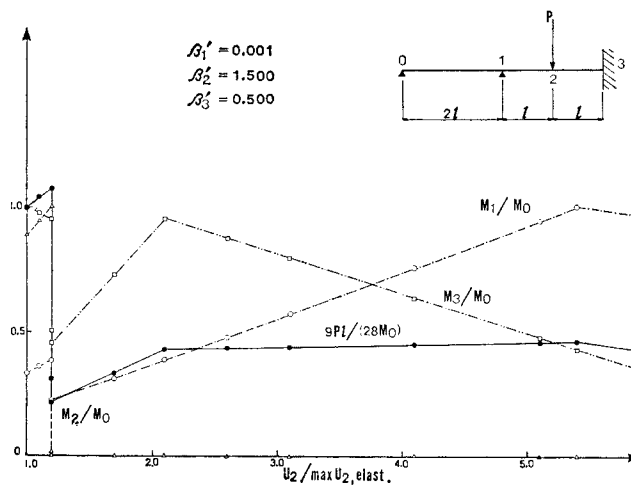


Fig. 9 Relations between bending moments or load and deflection for the problem including Case 1, Case 2, and Case 3.

M_i ($i=1, 2, 3$) shows bending moment at the point i , M_0 and β'_i ($=3I\beta_i/4EI$, EI is flexural rigidity) are yield moment and strain softening coefficient as defined in Fig. 2, u_2 is deflection at the loading point, and $\max u_{2, \text{elast.}}$ means the value of u_2 at the first yielding at the point 3.

Fig. 7 shows a problem where all of loading stages are included in only Case 1 mentioned before, and the problem of Fig. 8 includes both Case 1 and Case 2. It should be noted in these problems that the intensities of load and bending moments have a single value under a specified deflection, namely the uniqueness of solution is obtained in all loading stages.

On the other hand, the problem of Fig. 9 including Case 1, Case 2, and Case 3 no longer yields such a uniqueness of solution. Thus, a jumping of both the load and the bending moments under a constant value of deflection can be seen in this problem.

5. CONCLUDING REMARKS

A bending theory for elastic-plastic strain softening beams of statically indeterminacy is developed. A detailed procedure of numerical calculation based upon this theory is described through a simple example of a two-span beam. The procedure is applicable to a problem with large degrees of statically indeterminacy, except for the case where the nondiagonal elements of D in Prob. (13) include negative values. In such a case, some troublesome works are needed in connection with a nonconvex quadratic programming. Although this type of problem remains unsolved, it is believed that the procedure presented here gives an effective bridge to a general method.

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7. NOTATION

dW : strain energy increment.

$d\pi$: potential energy increment.

$d\delta$: vector of overall extremity-displacement increments in elements.

$\Delta\theta$: vector of overall rotation increments at strain-softening hinges.

$\varphi_{ia}, \varphi_{ib}$: element of $d\delta$, namely extremity-displacement increments at the left and the right ends of element i , respectively.

$\Delta\theta_j$: element of $\Delta\theta$, namely rotation increment at plastic strain-softening hinge i .

K_i : element stiffness matrix.

P_0 : normalized load vector.

dL : controlled displacement increment defined by Eq. (8).

F : lagrangian function defined by Eq. (12).

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和文要旨

ひずみ軟化を有する はりの曲げ解析

(園田恵一郎)

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コンクリート, 土や岩などの材料は, 圧縮荷重の作用下で, ひずみ軟化挙動を示すことはよく知られている。たとえば, コンクリートの円柱供試体の圧縮試験での応力・ひずみ曲線は約 0.25% のひずみを越えたところから圧壊の始まる約 0.35% の間で降下領域をもっている。このようなひずみ軟化挙動は物理的な不安定現象であるから, ひずみ軟化材料から成る構造物の応力および変形解析では, 解の安定性や唯一性の問題にしばしば遭遇する。

たとえば, 有限要素変位解析法における最終段階での方程式は次のように表わされる。

$$P = Ku \quad \dots\dots\dots (1)$$

ここに, P は規定された節点力または節点変位, u は未知の節点力または節点変位, K は全体剛性行列である。 K は構造形状と材料特性によって決まる行列であるから, ひずみ軟化特性にも影響を受ける。もし, K が正値 (positive definite) である場合は, 式 (1) の未知量 u は次式で唯一に決定できる。

$$u = K^{-1}P \quad \dots\dots\dots (2)$$

しかし, K が正値でない場合には, 式 (2) はもはや安定なつり合い解を与えない。すなわち, この場合は, 全体系のひずみエネルギーを与える関数が凸性を失い, 式 (2) の解が系のポテンシャルエネルギーを最小にする点に対応していないからである。

弾塑性・ひずみ軟化ばりのマトリックス変位法による曲げ解析においては, 節点力増分と節点変位増分の間に式 (1) と類似した関係が成立する。本論文は, モーメントと曲率の間に軟化挙動を有するはりに対して, 軟化の

程度と K の特性との関係および対応する安定なつり合い解の決定法について述べたものである。

最初に, たわみ増分の制御のもとでのひずみエネルギーの最小化原理について述べた。次に, はりの曲げ問題に対して, 塑性軟化ヒンジ (回転角の増加につれてモーメントが減少する特性をもつヒンジ) を導入し, マトリックス変位法による上述の最小化原理の定式化を試みた。得られた式中の未知量は塑性軟化ヒンジでの回転角増分, 分割された各要素の材端変位増分および荷重増分である。ラグランジュの未定係数法により自由変数である材端変位増分を消去し, 制約不等式をもつ回転角増分のみを変数にした問題に変換すると次の形を得る。

$$\left. \begin{aligned} \text{minimize } d\bar{W} &= -dLA^e\Delta\theta + \Delta\theta^e D \Delta\theta \\ \text{subject to } \Delta\theta &\geq 0 \end{aligned} \right\} \dots (3)$$

ここに, $\Delta\theta$ は塑性軟化ヒンジの回転角, dL はたわみ増分を与えるパラメーター, A, D は軟化率を含む係数行列である。

式 (3) の問題の解の決定には, 次の 3 つの場合に対し, それぞれ異なった取扱が必要である。

(1) D が正値 (positive definite) の場合

式 (3) は標準的な 2 次計画法 (Quadratic Programming) の問題になり, 周知の手法で唯一解が求められる。

(2) D が共正値 (copositive) の場合

$d\bar{W}$ は凸関数でないが, 領域 $\Delta\theta \geq 0$ では下に有界であるので, 式 (3) は解をもつ。

(3) D が正値でも共正値でもない場合

$d\bar{W}$ は下に有界でない。したがって, 式 (3) は解をもたない。すなわち, この場合は, 一部の塑性軟化ヒンジがひずみエネルギーを解放し, モーメントに抵抗しないヒンジになって, 不安定な系は新しい安定系に移行する。したがって, 一定のたわみのもとで, 荷重および曲げモーメント分布に飛移が起こる。

2 径間連続ばりの簡単な問題を通して上記の 3 つのケースに対する解の決定法を示し, 数値計算例により, 各ケースに対する荷重とたわみおよび曲げモーメントの関係を示した。