

## ANALYSIS OF CIRCULAR PLATE RESTING ON NON-TENSION FOUNDATION

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### 1. INTRODUCTION

Many works have been done on beams supported by Winkler's foundation, which is symbolised by a row of isolated elastic springs<sup>1)</sup>. This foundation is sometimes not the case for real applications. The ground for example does not move upward, therefore the Winkler's idea is not applicable to it if strictly speaking. The nontension foundation has been introduced to overcome this inaccuracy. It induces upward reaction force when it sinks, while it does not deflect upward together with beams on it. That is there is a separation between the foundation and the beams when they displace upwards.

Nien-chien Tsai and R. A. Westmann<sup>2)</sup>, and Y. Weitsman<sup>3)</sup> gave solutions for beams and circular plate of infinite radius on the nontension foundation. N. Kamiya derived a solution for a clamped circular plate<sup>4)</sup>.

In this report an analytical procedure will be carried out on a weightless circular plate subjected to a point load at the center and supported on non-tension Winkler's foundation.

### 2. METHOD OF ANALYSIS

#### (1) Governing Equations

Denote the displacement of the middle surface of the plate by  $w$  ( $w$  positive downward) and the radial distance by  $r$ . The radius of contact is denoted by  $r_0$  (unknown) where the plate separates from the elastic foundation.

The governing equations of axisymmetric plates on the nontension foundation are<sup>5)</sup>:

at  $0 < r < r_0$  and  $w > 0$ ;

$$D \nabla_r^2 \nabla_r^2 w_1 + k w_1 = 0 \quad \dots\dots\dots (1)$$

at  $r > r_0$  and  $w < 0$ ;

$$\nabla_r^2 \nabla_r^2 w_2 = 0 \quad \dots\dots\dots (2)$$

In Eqs. (1) and (2)  $\nabla_r^2$  denotes Laplacian operator for 1-dimension and

$$\nabla_r^2 = \frac{1}{r} \frac{d}{dr} \left( r \frac{d}{dr} \right) \quad \dots\dots\dots (3)$$

$D$  is the flexural rigidity of the plate,  $k$  is the modulus of the foundation and the subscripts 1 and 2 refer to the contact and the no-contact regions respectively.

Denote

$$l^4 = \frac{D}{k} \quad \dots\dots\dots (4)$$

Introduce the dimensionless quantities  $x, x_0$  and  $\gamma$  defined by

$$x = r/l, \quad x_0 = r_0/l, \quad \gamma = a/l \quad \dots\dots\dots (5 \cdot a, b, c)$$

where  $a$  denotes radius of the plate.

The general solutions of Eqs. (1) and (2) are respectively<sup>5),6)</sup>:

$$w_1(x) = C_0 \text{ber } x + E_0 \text{bei } x + F_0 \text{ker } x + G_0 \text{kei } x \quad \dots\dots\dots (6)$$

$$w_2(x) = A + Bx^2 + C \ln x + Ex^2 \ln x \quad \dots\dots\dots (7)$$

where  $\text{ber } x, \text{bei } x, \text{ker } x$  and  $\text{kei } x$  are Kelvin functions of order 0.

Because the deflection must be finite at the origin,  $F_0$  is set to equal zero since  $\text{ker } x$  becomes singular at the origin<sup>7)</sup>. Thus Eq. (6) becomes

$$w_1(x) = C_0 \text{ber } x + E_0 \text{bei } x + G_0 \text{kei } x \quad \dots\dots\dots (6 \cdot a)$$

#### (2) Boundary and Continuity Conditions

For the contact and no-contact domains of the plate, the boundary conditions and continuity conditions are respectively:

$$(M_{r2})_{x=\gamma} = 0 \quad \dots\dots\dots (8)$$

$$(Q_{r2})_{x=\gamma} = 0 \quad \dots\dots\dots (9)$$

$$(W_1)_{x=x_0} = 0 \quad \dots\dots\dots (10)$$

$$(w_2)_{x=x_0} = 0 \quad \dots\dots\dots (11)$$

$$(Q_{r1})_{x=x_0} = 0 \quad \dots\dots\dots (12)$$

$$(w_1')_{x=x_0} = (w_2')_{x=x_0} \quad \dots\dots\dots (13)$$

$$(M_{r1})_{x=x_0} = (M_{r2})_{x=x_0} \quad \dots\dots\dots (14)$$

The equilibrium of transverse forces requires that

$$\left[ \int_0^{2\pi} Q_{rr} d\theta \right]_{r=\epsilon} + P = 0 \quad \dots\dots\dots (15)$$

where  $\epsilon$  is an infinitesimal positive value.

$M_r$  and  $Q_r$  are respectively the radial bending

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moment and the radial shearing force per unit width of the section and are

$$M_r = -\frac{D}{l^2} \left[ \frac{d^2 w}{dx^2} + \nu \frac{1}{x} \frac{dw}{dx} \right] \dots\dots\dots(16)$$

$$Q_r = -\frac{d}{l^3} \frac{d}{dx} (V r^2 w) \dots\dots\dots(17)$$

Other stress resultant besides those defined in Eqs. (16) and (17) is

$$M_\theta = -\frac{D}{l^2} \left( \frac{1}{x} \frac{dw}{dx} + \nu \frac{d^2 w}{dx^2} \right) \dots\dots\dots(18)$$

The stress resultants as defined in Eqs. (16) to (18) can be evaluated with the help of the following expressions for the derivatives of Kelvin functions<sup>7)</sup> which appear in Eq. (6).

$$\frac{d}{dx} f_m = \frac{m}{x} f_m + \frac{1}{\sqrt{2}} (f_{m+1} + g_{m+1}), \quad x > 0 \dots\dots\dots(19)$$

$$\frac{d^2}{dx^2} f_m = \frac{m}{x^2} (m-1) f_m - \frac{1}{x \sqrt{2}} (f_{m+1} + g_{m+1}) - g_m, \quad x > 0 \dots\dots\dots(20)$$

$$\begin{aligned} \frac{d^3}{dx^3} f_m &= \frac{m(m-1)(m-2)}{x^3} f_m \\ &+ \frac{1}{x^2 \sqrt{2}} (m^2 + 2 + x^2) f_{m+1} \\ &+ \frac{1}{x^2 \sqrt{2}} (m^2 + 2 - x^2) g_{m+1} \\ &- \frac{m-1}{x} g_m, \quad x > 0 \dots\dots\dots(21) \end{aligned}$$

where  $f_m$  and  $g_m$  stand for one of the following four sets of Kelvin functions of  $m$ th order

$$\begin{aligned} \left. \begin{aligned} f_m &= \text{ber}_m x \\ g_m &= \text{bei}_m x \end{aligned} \right\} \dots\dots\dots(22 \cdot a, b) \\ \left. \begin{aligned} f_m &= \text{kei}_m x \\ g_m &= -\text{ber}_m x \end{aligned} \right\} \\ \left. \begin{aligned} f_m &= \text{ker}_m x \\ g_m &= \text{kei}_m x \end{aligned} \right\} \\ \left. \begin{aligned} f_m &= \text{kei}_m x \\ g_m &= -\text{ker}_m x \end{aligned} \right\} \dots\dots\dots(22 \cdot c, d) \end{aligned}$$

The behaviour of each Kelvin function and its contribution to the stress resultants are shown in Appendix.

**3. SOLUTIONS**

Substituting Eq. (17) into Eq. (15) and knowing that as  $x$  tends to zero,  $\text{kei } x$  is the only term in Eq. (6·a) which contributes to the values of  $Q_r$  and this leads to

$$G_0 = -\frac{Pl^3}{D} \frac{1}{\int_0^{2\pi} \left[ -\frac{1}{\sqrt{2}} (\text{ker}_1 x + \text{kei}_1 x) \right] r d\theta} \dots\dots\dots(23)$$

after introducing a integrated denominator,

$$G_0 = -\frac{Pl^2}{2\pi D} \dots\dots\dots(24)$$

At  $x = \gamma$ , the radial shear vanishes, thus the Eq. (9) leads to

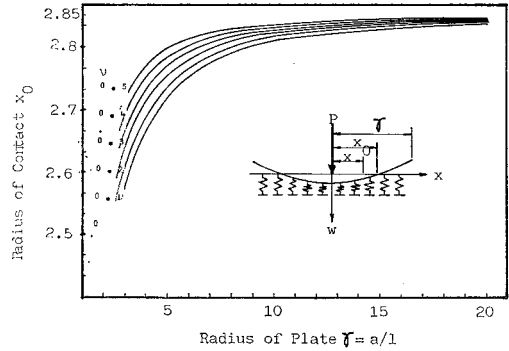


Fig. 1 Radius of Contact  $x_0$ (Solutions of Eq. (36)).

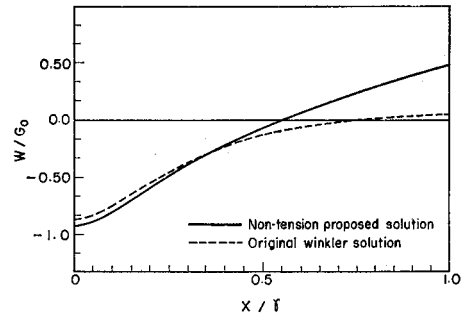


Fig. 2 Deflections of plates under a concentrated load at the center of plates.  $a$ =radius of plate,  $D$ =flexural rigidity of plate,  $k$ =modulus of foundation,  $P$ =concentrated load,  $r$ =radial coordinate,  $w$ =deflection,  $l = \sqrt[4]{D/k}$ ,  $G_0 = -Pl^2/2\pi D$ ,  $x = r/l$ ,  $x/\gamma = r/a$ . ( $\gamma = 5$ ,  $\nu = 0.2$ )

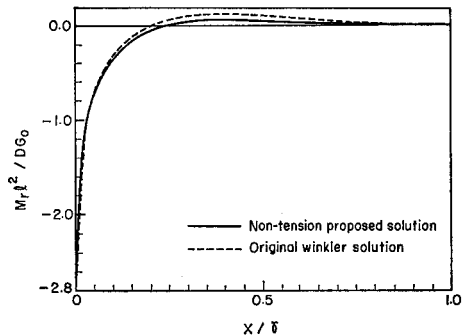


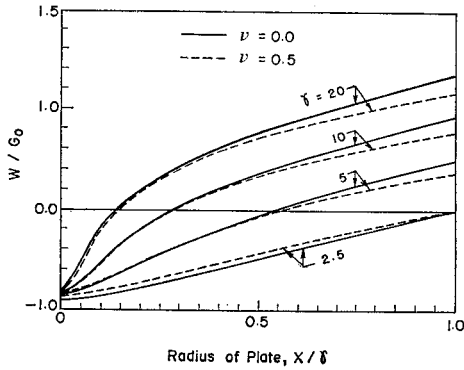
Fig. 3 Moment of plates.  $M$ =radial moment,  $a$ =radius of plate,  $D$ =flexural rigidity of plate,  $k$ =modulus of foundation,  $P$ =concentrated load,  $r$ =radial coordinate,  $l = \sqrt[4]{D/k}$ ,  $G_0 = -Pl^2/2\pi D$ ,  $x = r/l$ ,  $x/\gamma = r/a$ . ( $\gamma = 5$ ,  $\nu = 0.2$ )

$$E = 0 \dots\dots\dots(25)$$

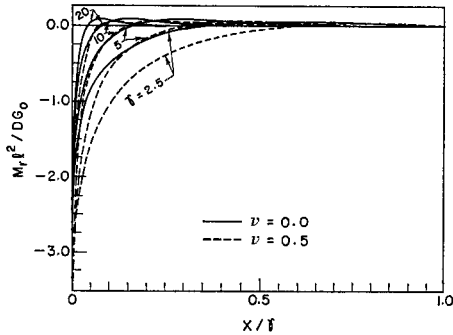
From the other conditions, the arbitrary constants  $C_0$ ,  $E_0$ ,  $A$ ,  $B$  and  $C$  are respectively,

$$C_0 = -\frac{G_0}{\text{ber } x_0} [\text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0] \dots\dots\dots(26)$$

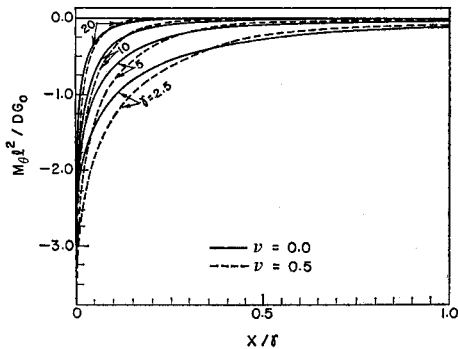
$$E_0 = -G_0 \zeta_1(x_0) \dots\dots\dots(27)$$



**Fig. 4** Deflection of plate  
 $\nu$  = Poisson's ratio,  $a$  = radius of plate,  $D$  = flexural rigidity of plate,  $k$  = modulus of foundation,  $P$  = concentrated load,  $r$  = radial coordinate,  $w$  = deflection,  $l = \sqrt[4]{D/k}$ ,  $G_0 = -Pl^2/2\pi D$ ,  $x = r/l$ ,  $x/l = r/a$ .



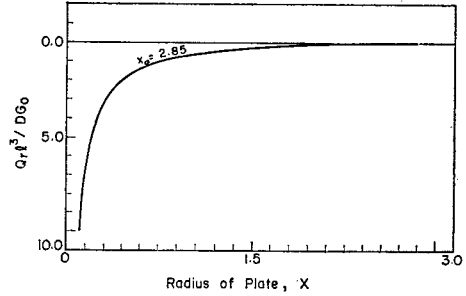
**Fig. 5** Distribution of radial bending moment. Refer to notations of Fig. 3.



**Fig. 6** Distribution of transverse bending moment  $M_\theta$ . Refer to notations of Fig. 3.

$$C = -\frac{G_0 x_0}{2} \frac{\text{kei } x_0 - \zeta_1(x_0) \text{bei } x_0}{\text{ber } x_0} \left[ \left\{ \sqrt{2} R_1(x_0) + x_0 \text{bei } x_0 \right\} + \zeta_1(x_0) \left\{ \sqrt{2} S_1(x_0) - x_0 \text{ber } x_0 \right\} + \left\{ \sqrt{2} U_1(x_0) - x_0 \text{ker } x_0 \right\} \right] \dots (28)$$

$$B = \frac{-G_0}{2x_0} \left[ \frac{R_1(x_0)}{\sqrt{2}} \frac{\text{kei } x_0 - \zeta_1(x_0) \text{bei } x_0}{\text{ber } x_0} \right]$$



**Fig. 7** Distribution of radial shearing force  $Q_r$ . Refer to notations of Fig. 3.  $X_0 = 2.85$  for an ideal case of  $\nu = 0.5$  and  $\gamma = \infty$ .

$$+ \zeta_1(x_0) \frac{S_1(x_0)}{\sqrt{2}} + \frac{U_1(x_0)}{\sqrt{2}} \frac{C}{x_0} \dots (29)$$

$$A = -Bx_0^2 - C \ln x_0 \dots (30)$$

where

$$R_1(x_0) = \text{ber}_1 x_0 + \text{bei}_1 x_0 \dots (31)$$

$$S_1(x_0) = \text{bei}_1 x_0 - \text{ber}_1 x_0 \dots (32)$$

$$T_1(x_0) = \text{ker}_1 x_0 + \text{kei}_1 x_0 \dots (33)$$

$$U_1(x_0) = \text{kei}_1 x_0 - \text{ker}_1 x_0 \dots (34)$$

$$\zeta_1(x_0) = \frac{\text{kei } x_0 S_1(x_0) + \text{ber } x_0 T_1(x_0)}{\text{bei } x_0 S_1(x_0) + \text{ber } x_0 R_1(x_0)} \dots (35)$$

The condition for unknown  $x_0$  is derived,

$$\begin{aligned} & -\frac{1}{\text{ber } x_0} [\text{kei } x_0 - \zeta_1(x_0) \text{bei } x_0] \\ & \cdot \left[ \text{bei } x_0 + \frac{1}{\gamma^2} \frac{(1-\nu)}{(1+\nu)} x_0 \left\{ \sqrt{2} R_1(x_0) + x_0 \text{bei } x_0 \right\} - \zeta_1(x_0) \left[ \text{ber } x_0 - \frac{1}{\gamma^2} \frac{(1-\nu)}{(1+\nu)} x_0 \right. \right. \\ & \cdot \left. \left. \left\{ \sqrt{2} S_1(x_0) - x_0 \text{ber } x_0 \right\} \right] + \left[ \text{ker } x_0 - \frac{1}{\gamma^2} \frac{(1-\nu)}{(1+\nu)} x_0 \left\{ \sqrt{2} U_1(x_0) - x_0 \text{ker } x_0 \right\} \right] = 0 \end{aligned} \dots (36)$$

From Eq. (36), the weightless plate lifts up at a distance which is independent of the magnitude of  $P$ . The unknown  $X_0$  is derived numerically by a trial and error method for given  $\gamma$  and  $\nu$  values. The contact radius  $X_0$  increases as  $\gamma$  or  $\nu$  increases.

Substituting Eqs. (28), (29) and Eq. (30) into Eq. (7) leads to

$$\begin{aligned} \frac{w_2}{G_0} = & -\frac{x_0}{2} \left[ \frac{(x^2 - x_0^2)}{2\gamma^2} \frac{(1-\nu)}{(1+\nu)} + \ln \frac{x}{x_0} \right] \\ & \cdot \left[ \frac{\text{kei } x_0 - \zeta_1(x_0) \text{bei } x_0}{\text{ber } x_0} \left\{ \sqrt{2} R_1(x_0) + x_0 \text{bei } x_0 \right\} + \zeta_1(x_0) \left\{ \sqrt{2} S_1(x_0) - x_0 \text{ber } x_0 \right\} + \left\{ \sqrt{2} U_1(x_0) - x_0 \text{ker } x_0 \right\} \right] \dots (37) \end{aligned}$$

Substituting Eqs. (26) and (27) into Eq. (6a) leads to

$$\frac{w_1}{G_0} = -\frac{1}{\text{ber } x_0} [\text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0] \text{ ber } x - \zeta_1(x_0) \text{ bei } x + \text{kei } x \dots\dots\dots(38)$$

Finally substituting Eqs. (37) and (38) into Eqs. (16) to (18) leads to the equation of stress resultants as follows:

$$M_{r1} \frac{l^2}{DG_0} = -\frac{1}{\text{ber } x_0} (\text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0) \cdot \left\{ \frac{1-\nu}{x \sqrt{2}} R_1(x_0) + \text{bei } x \right\} - \zeta_1(x_0) \left\{ \frac{(1-\nu)}{x \sqrt{2}} S_1(x_0) - \text{ber } x \right\} + \left\{ \frac{1-\nu}{x \sqrt{2}} U_1(x) - \text{ker } x \right\} \dots\dots(39)$$

$$M_{r2} \frac{l^2}{DG_0} = -\frac{x_0}{2} \left[ \frac{1-\nu}{1+\nu} \frac{1}{\gamma^2} - \frac{1}{x^2} + \nu \left\{ \frac{1-\nu}{1+\nu} \frac{1}{\gamma^2} + \frac{1}{x^2} \right\} \right] \left[ -\frac{1}{\text{ber } x_0} \cdot \{ \text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0 \} + x_0 \text{ bei } x_0 \right] - \zeta_1(x_0) \{ \sqrt{2} S_1(x_0) - x_0 \text{ ber } x_0 \} + \{ \sqrt{2} U_1(x_0) - x_0 \text{ ker } x_0 \} \dots\dots(40)$$

$$M_{\theta 1} \frac{l^2}{DG_0} = -\frac{1}{\text{ber } x_0} (\text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0) \cdot \left[ \nu \text{ bei } x - \frac{1-\nu}{x \sqrt{2}} R_1(x) \right]$$

$$-\zeta_1(x_0) \left[ -\nu \text{ ber } x - \frac{1-\nu}{x \sqrt{2}} S_1(x) \right] - \nu \text{ ker } x - \frac{1-\nu}{x \sqrt{2}} U_1(x) \dots\dots(41)$$

$$M_{\theta 2} \frac{l^2}{DG_0} = -\frac{x_0}{2} \left[ \frac{1-\nu}{1+\nu} \frac{1}{\gamma^2} + \frac{1}{x^2} + \nu \left\{ \frac{1-\nu}{1+\nu} \frac{1}{\gamma^2} - \frac{1}{x^2} \right\} \right] \left[ -\frac{1}{\text{ber } x_0} \cdot \{ \text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0 \} \cdot \{ \sqrt{2} R_1(x_0) + x_0 \text{ bei } x_0 \} - \zeta_1(x_0) \{ \sqrt{2} S_1(x_0) - x_0 \text{ ber } x_0 \} + \{ \sqrt{2} U_1(x_0) - x_0 \text{ ker } x_0 \} \right] \dots(42)$$

$$\frac{Q_{r1}}{G_0} = -\frac{S_1(x_0)}{\sqrt{2} \text{ber } x_0} (\text{kei } x_0 - \zeta_1(x_0) \text{ bei } x_0) + \frac{\zeta_1(x_0)}{\sqrt{2}} R_1(x) - \frac{T_1(x)}{\sqrt{2}} \dots\dots(43)$$

Some numerical results are shown in figures.

**4. ACKNOWLEDGEMENTS**

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**APPENDIX**

The contribution of the Kelvin functions to the deflection and the stress results are given below:

Item	Expression		
	ber <i>x</i>	bei <i>x</i>	kei <i>x</i>
$\frac{l^2}{D} M_r$	$\frac{1-\nu}{x \sqrt{2}} (\text{ber}_1 x + \text{bei}_1 x) - \text{ber } x$	$\frac{1-\nu}{x \sqrt{2}} (\text{bei}_1 x - \text{ber}_1 x) - \text{ber } x$	$\frac{1-\nu}{x \sqrt{2}} (\text{kei}_1 x - \text{ker}_1 x) - \text{ker } x$
$\frac{l^2}{D} M_\theta$	$\nu \text{ bei } x - \frac{1-\nu}{x \sqrt{2}} (\text{ber}_1 x + \text{bei}_1 x)$	$-\nu \text{ ber } x - \frac{1-\nu}{x \sqrt{2}} (\text{bei}_1 x - \text{ber}_1 x)$	$-\nu \text{ kei } x - \frac{1-\nu}{x \sqrt{2}} (\text{kei}_1 x - \text{ker}_1 x)$
$\frac{l^2}{D} Q_r$	$\frac{1}{\sqrt{2}} (\text{bei}_1 x - \text{ber}_1 x)$	$-\frac{1}{\sqrt{2}} (\text{bei}_1 x + \text{ber}_1 x)$	$-\frac{1}{\sqrt{2}} (\text{ker}_1 x + \text{kei}_1 x)$

Subscripts 1 to bei and ber functions denote the functions of order 1

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